

Fig. 11. Out-of-plane displacements: a) minimum load configuration (pre-buckling - 1.8 kN); b) maximum load configuration (post-buckling - 18 kN).

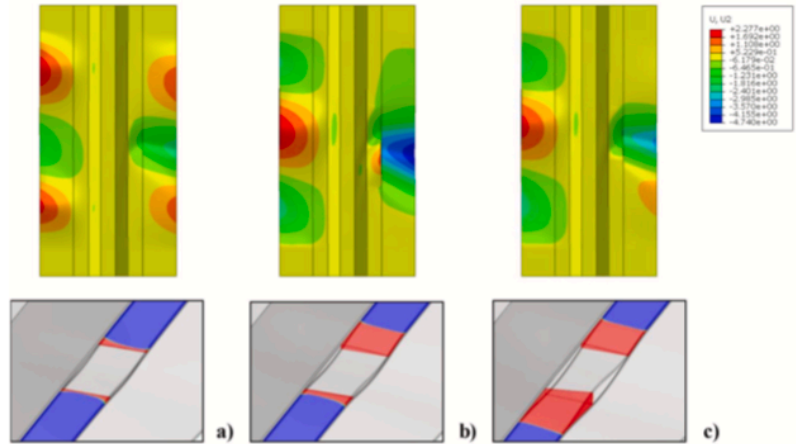


Fig. 12. Out-of-plane displacements at maximum load and delamination front after: a) 5000 cycles; b) 7800 cycles; c) 8100 cycles.

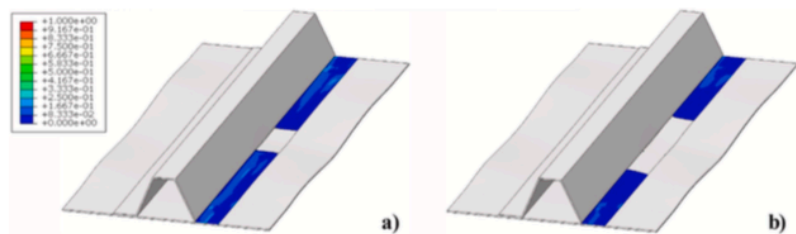


Fig. 13. Distribution of the state variable related to the local stress ratio at: a) first cycle; b) 10,000 cycles.



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