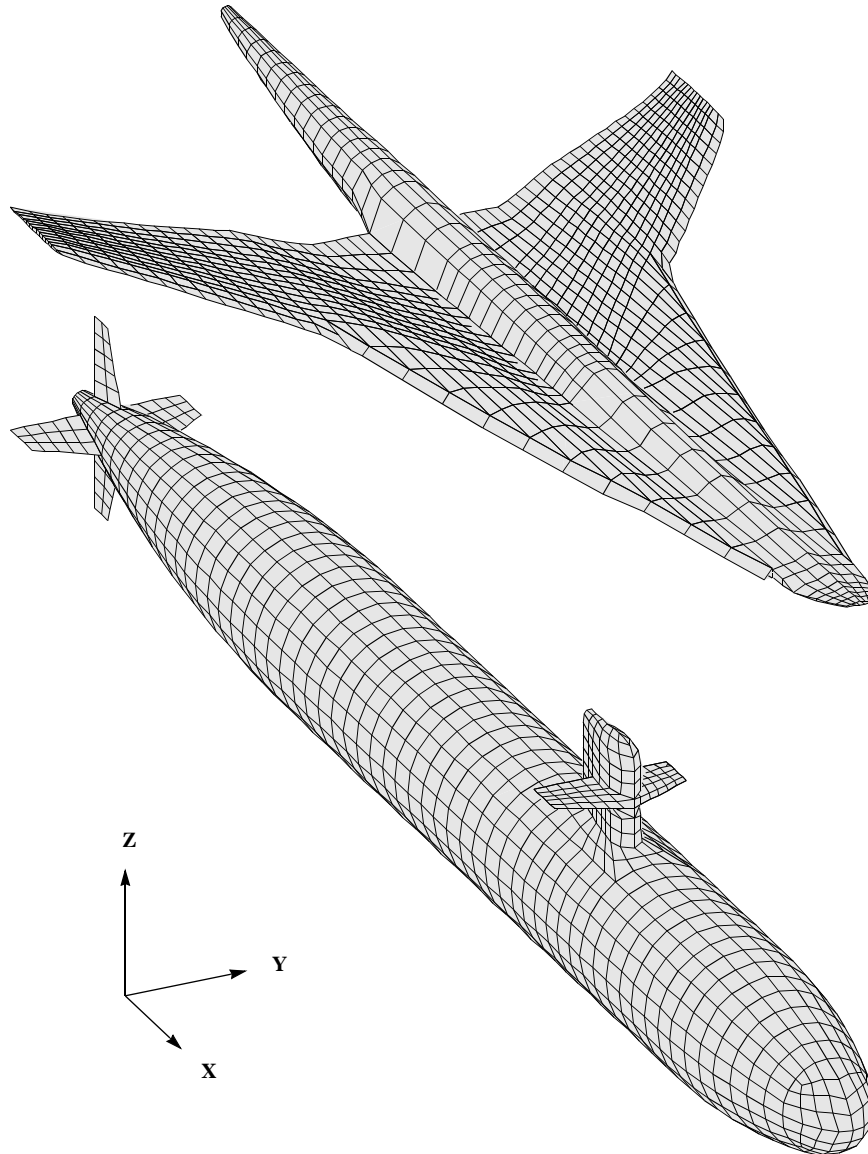


# STAGS

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## *ST*ructural *A*nalysis of *G*eneral *S*hells



**Lockheed-Martin Advanced Technology Center**

## ANALYSIS OPTIONS

- **static**
  - geometrically linear or nonlinear
  - linear elasticity or plasticity
- **buckling**
  - bifurcation based upon either a linear or a nonlinear stress state
- **branch on bifurcation**
  - from a primary solution path to a secondary solution path
- **vibration**
  - based upon either a linear or a nonlinear stress state
- **transient**
  - geometrically linear or nonlinear
  - linear elasticity or plasticity

## ELEMENTS

- **shells**
  - 3-node thin triangle
  - 4-node thin quadrilateral
  - 9-node thick quadrilateral
  - transition elements for mesh refinement
- **beams**
  - 2-node Timoshenko beam
- **nonlinear spring**
  - general, velocity-dependent, force-displacement behavior
- **special elements**
  - hyperelastic fastener element
  - moving-plane boundary
- **contact elements**
  - general point/surface contact
  - surface/surface pad contact
- **solid elements**

## SOLUTION STRATEGIES

- **load/displacement control**
- **Riks arc-length technique**
  - traversal of limit points into post-buckling regimes
- **equivalence transformation**
  - switch from the primary solution path to a secondary solution path
- **restart**
  - from any previous solution
  - switch analysis option; e.g., from static to transient
  - change loads, boundary conditions

## FABRICATION

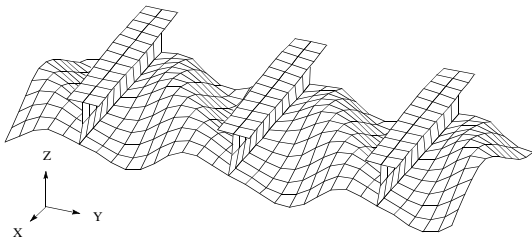
- **shell walls**
  - laminate composite
  - corrugation-stiffened
  - general—stiffness matrix
  - smeared stiffeners
- **beam cross sections**
  - cross-section properties
  - sub-element properties
  - rectangular sub-elements
  - general—stiffness matrix
- **materials**
  - linear elastic
  - elastic/plastic with strain hardening
  - orthotropic properties
  - power-law creep

## BOUNDARY CONDITIONS

- **standard B.C. types**
  - simple, (anti)symmetry, fixed, etc.
- **specified displacements**
- **displacement constraints**
  - SPC—single-point constraints
  - MPC—multi-point constraints; linear and nonlinear
  - Lagrange constraints—linear equations
- **initial conditions**
  - initial displ./velocity for transient analysis

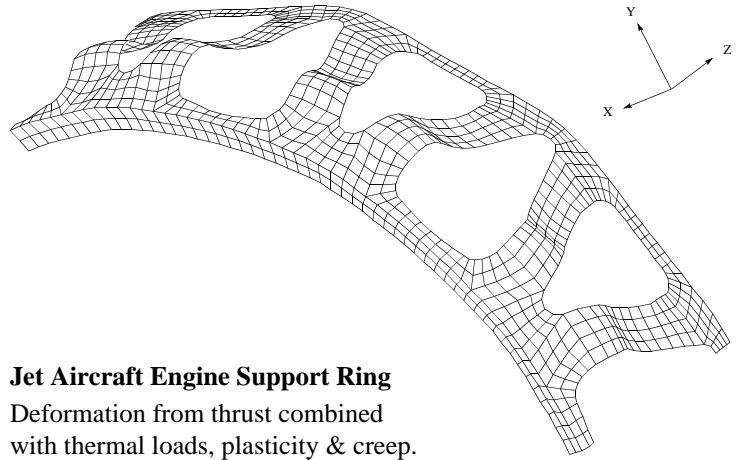
## LOADS

- **mechanical loads**
  - point loads, line loads, surface traction
  - ‘live’ pressure—remains normal to the deformed surface
- **thermal loads**
  - arbitrary temperature distribution throughout structure



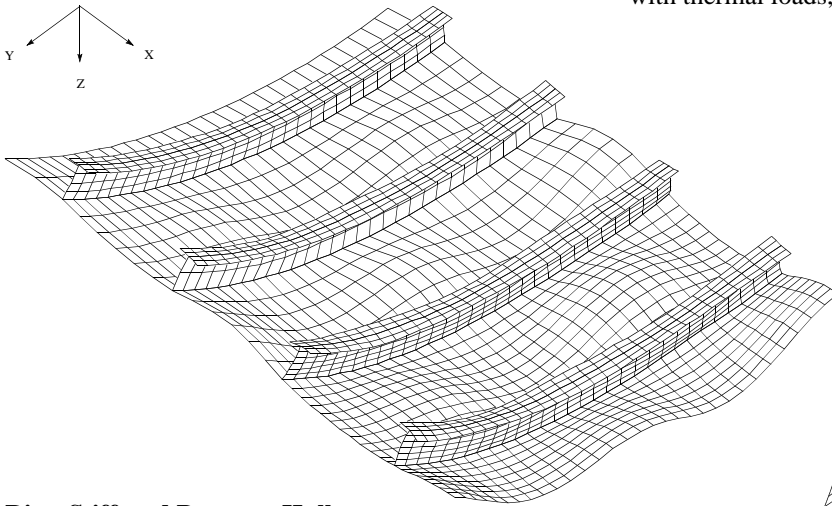
**Tee-Stiffened Panel**

Highly-stressed, post-buckled shape produced by normal pressure and shear.



**Jet Aircraft Engine Support Ring**

Deformation from thrust combined with thermal loads, plasticity & creep.

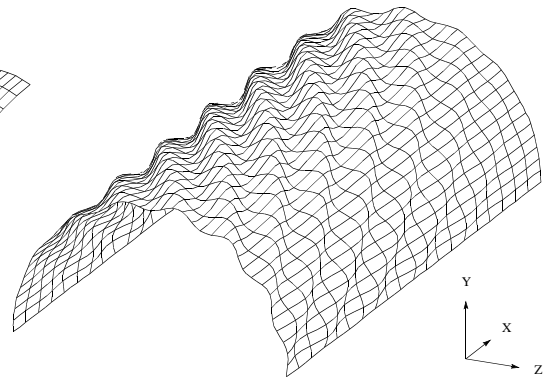


**Ring-Stiffened Pressure Hull**

Plastic collapse of imperfect cylinder.

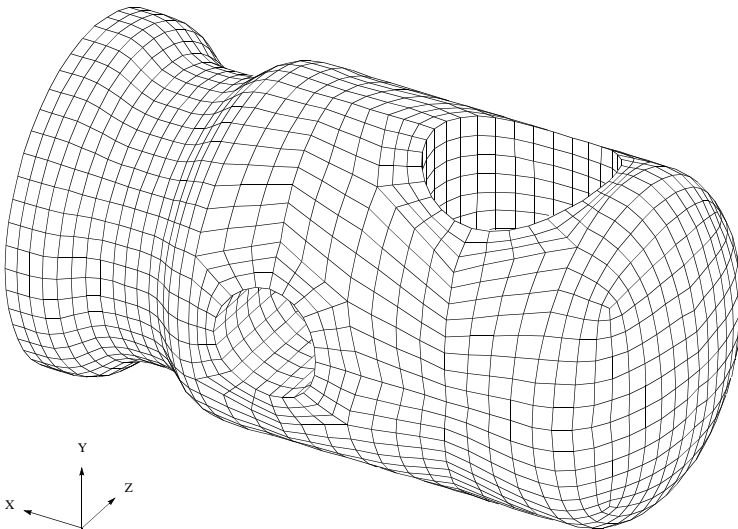
**Composite Cylinder**

Buckling of asymmetrical fabrication.



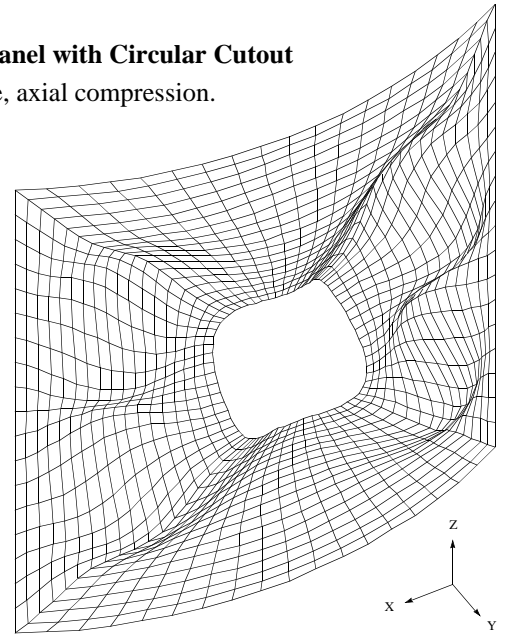
**Complex Titanium Pressure Vessel**

Intersecting cylinders, ellipsoidal end cap, toroidal radius reduction.



**Cylindrical Panel with Circular Cutout**

Buckled shape, axial compression.



## MODEL GENERATION & RESULTS PRESENTATION

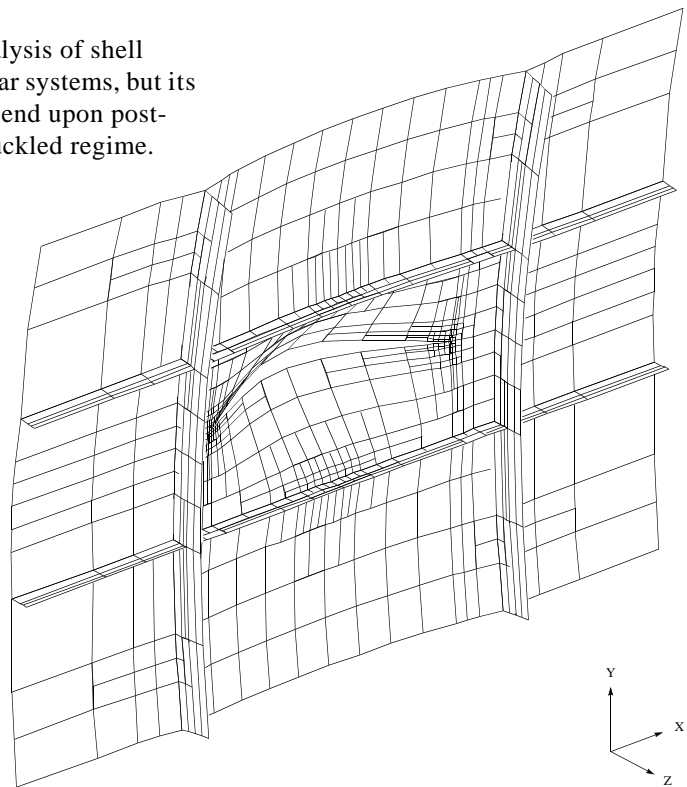
- **shell units** – substructures defined using a large catalog of quadric surfaces  
– automatic generation, discretization, and assembly of substructures  
– complex models built with minimal user input
- **element units** – arbitrary assemblages of nodes and elements  
– most useful for inputting model data from a preprocessor
- **STAPL** – **STAGS *Plotting program***—post-processor for generating model plots and  
– plots showing displacement, stress and other results obtained by **STAGS**
- **STAR** – **STAGS Access Routines**—high-level read/write database access  
– facilitates user-development of translators
- **PATRAN** – forward & backward PATRAN translators (utilizing **STAR**) included

*STAGS* is a finite element program for general-purpose analysis of shell structures. *STAGS* can be applied efficiently to routine linear systems, but its forte is the analysis of complex, nonlinear systems that depend upon post-buckling strength and require analysis well into the post-buckled regime.

*STAGS* is routinely used for pre- and post-test verification of complex systems — especially those that are sensitive to initial geometric imperfections, which can be defined in *STAGS* with ease and flexibility.

*STAGS* has been under continuous development since the 1960's by Lockheed's Research and Development Division and Lockheed-Martin's Advanced Technology Center, whose goals have always been to achieve the best shell-stability code available anywhere.

*STAGS* has a world-wide distribution and is used extensively throughout government, industry and academia. Current *STAGS* research topics include crack-growth simulation and adaptive mesh refinement, illustrated in the pressurized aircraft fuselage model shown at the right.



### MACHINES

UNIX Workstations

Sun  
DEC  
SGI  
IBM  
HP

Supercomputer

CRAY

PC, Macintosh—under development

### INFORMATION & ORDERS

COSMIC

NASA's Software Technology Transfer Center  
The University of Georgia  
382 East Broad St.  
Athens, GA 30602-4272 USA

Phone: 706-542-3265

Fax: 706-542-4807

service@cossack.cosmic.uga.edu