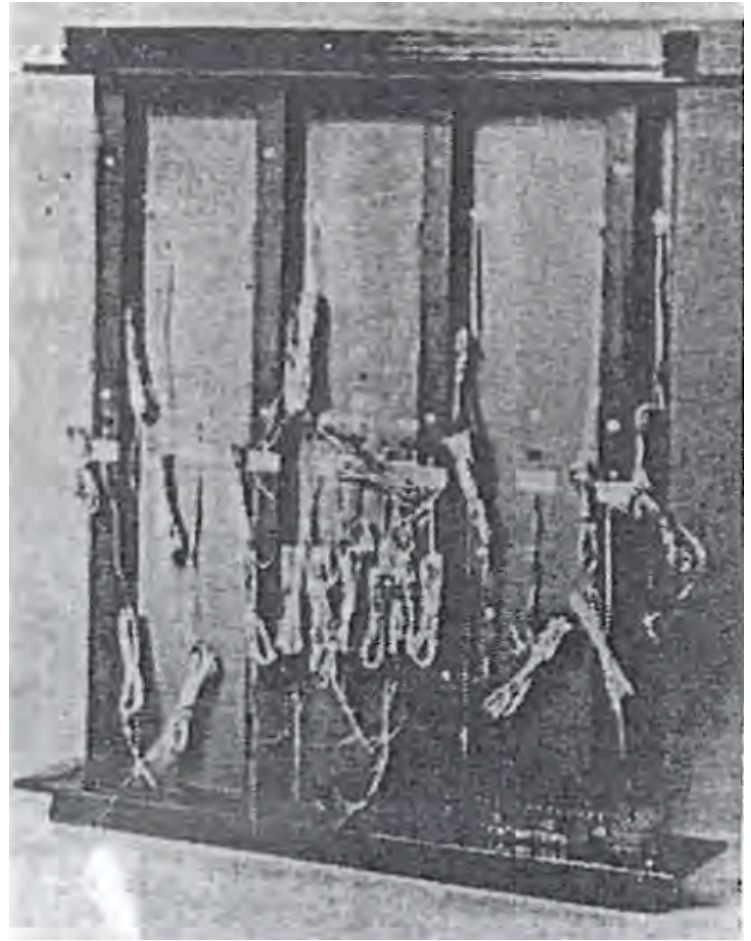




Dr. Norman F. Knight, Jr.



From: Norman F. Knight, Jr., "Finite Element Techniques for Nonlinear Postbuckling and Collapse of Elastic Structures", Chapter 1 in *Structural Dynamic Systems: Computational Techniques and Optimization*, Edited by Cornelius T. Leondes, Gordon and Breach Science Publishers, 1998

NASA Langley Research Center, Hampton, Virginia

Selected Publications:

Ahmed K. Noor and Norman F. Knight, Jr. (George Washington University Center at NASA Langley Research Center, Hampton, VA, U.S.A), "Nonlinear dynamic analysis of curved beams", *Computer Methods in Applied Mechanics and Engineering*, Vol. 23, No. 2, August 1980, pp. 225-251, doi:10.1016/0045-7825(80)90095-X

J. H. Starnes, Jr., N. F. Knight, Jr. and M. Rouse, Postbuckling behavior of selected flat stiffened graphite-epoxy panels loaded in compression, AIAA Paper 82-0777, presented at AIAA 23rd Structures, Structural Dynamics, and Materials Conference, New Orleans, May, 1982. See also, AIAA J., 23, (8) (1985) pp.1236-1246.

Knight, Norman F, Jr and Starnes, James H, Jr, "Postbuckling behavior of axially compressed graphite-epoxy cylindrical panels with circular holes", American Society of Mechanical Engineers, Joint Pressure Vessels and Piping/Applied Mechanics Conference, San Antonio, TX; United States; 17-21 June 1984

Knight, N F, Jr, Starnes, J H, Jr And Waters, W A, Jr, "Postbuckling behavior of selected graphite-epoxy cylindrical panels loaded in axial compression", AIAA Structures, Structural Dynamics and Materials Conference, 27th, San Antonio, TX; USA; 19-21 May 1986. pp. 142-158. 1986

Knight N. F. Jr. and Starnes J. H. Jr., Postbuckling behavior of selected curved stiffened graphite- epoxy panels loaded in axial compression, AIAA Journal, Vol. 26, No. 3, 1988, pp.344-352.

Knight, N. F., Jr., Gillian, R. E., McCleary, S. L., Lotts, C. G., Poole, E. L., Overman, A. L., and Macy, S. C.: CSM Testbed Development and Large-Scale Structural Applications. NASA TM-4072, 1989.

Knight, N. F., Jr., Greene, W. H., and Stroud, W. J.: Nonlinear Response of a Blade-Stiffened Graphite-Epoxy Panel with a Discontinuous Stiffener. Computational Methods for Structural Mechanics and Dynamics, Part 1, W. Jefferson Stroud, Jerrold M. Housner, John A. Tanner, and Robert J Hayduk (Compilers), NASA CP-3034 Part 1, 1989, pp. 51-65.

Knight, N. F., Jr., McCleary, S. L., Macy, S. C., and Aminpour, M. A.: Large-Scale Structural Analysis: The Structural Analyst, The CSM Testbed, and The NAS System. NASA TM-100643, March 1989.

Ransom, J. B., and Knight, N. F., Jr., "Global/Local Stress Analysis of Composite Panels," Computers and Structures, Vol. 37, No. 4, 1990, pp. 375-395.

Engelstad, S. P., Reddy, J. N., and Knight, N. F.: Postbuckling Response and Failure Prediction of Graphite-Epoxy Plates Loaded in Compression. AIAA Journal, Vol. 30, No. 8, August 1992, pp. 2106-2113.

Norman F. Knight Jr (Department of Aerospace Engineering, Old Dominion University, Norfolk, Virginia), "Factors influencing nonlinear static response prediction and test-analysis correlation for composite panels", Composite Structures, Vol. 29, No. 1, 1994, pp. 13-25, doi:10.1016/0263-8223(94)90033-7

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"Assessment of structural analysis technology for static collapse of elastic cylindrical shells", Finite Elements in Analysis and Design, Vol. 18, No. 4, January 1995, pp. 403-431, doi:10.1016/0168-874X(94)00066-O

Jaunky, N., Knight, N.F., Ambur, D.R., "Buckling of arbitrary quadrilateral anisotropic plates," AIAA Journal, 1995, 3, 938-44.

Jaunky, N., Knight, N.F., Ambur, D.R., "Buckling analysis of general triangular anisotropic plates using polynomials," AIAA Journal, 1995, 33, 2414-7.

Rengarajan, G. (1), Aminpour, M. A. (2) and Knight, N. F. (3),
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(3) Department of Aerospace Eng., Old Dominion University, Norfolk, VA 23529-0247, U.S.A.
"Improved assumed-stress hybrid shell element with drilling degrees of freedom for linear stress, buckling and free vibration analyses", International Journal for Numerical Methods in Engineering, 1995, 38: 1917-1943.
doi: 10.1002/nme.1620381108

Knight, Norman F, Jr (NASA Langley Research Center, Hampton, VA 23681-0001, USA), "Assumed--stress hybrid elements with drilling degrees of freedom for nonlinear analysis of composite structures (Final Report, ending Dec. 31, 1995)

N.F. Knight Jr. The Raasch challenge for shell elements. In Proc. 37th AIAA/ ASME/ASCE/AHS/ASC Structures, Structural Dynamics and Material Conference, CP. 962, pages 450-460, Salt Lake City, UT, 1996.

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“Formulation of an improved smeared stiffener theory for buckling analysis of grid-stiffened composite panels”, Composites Part B: Engineering, Vol. 27, No. 5, 1996, pp. 519-526, doi:10.1016/1359-8368(96)00032-7

Knight, Norman F, Jr and Nemeth, Michael P (editors), “Stability Analysis of Plates and Shells”, NASA no. 19980019011. In: AIAA/ASME/ASCE/AHS/ASC Structures, Structural Dynamics, and Materials Conference

Norman F. Knight, Jr and Yunqian Qi (Department of Aerospace Engineering, Old Dominion University, Norfolk, VA 23529-0247, USA), “On a consistent first-order shear-deformation theory for laminated plates”, Composites Part B: Engineering, Vol. 28, No. 4, 1997, pp. 397-405, doi:10.1016/S1359-8368(96)00058-3

Norman F. Knight, Jr and Yunqian Qi (Department of Aerospace Engineering, Old Dominion University, Norfolk, Virginia 23529-0247, U.S.A.), “Restatement of first-order shear-deformation theory for laminated plates”, International Journal of Solids and Structures, Vol. 34, No. 4, February 1997, pp. 481-492, doi:10.1016/S0020-7683(96)00032-7

Knight, Norman F, Jr and Starnes, James H, Jr, “Developments in Cylindrical Shell Stability Analysis”, NASA no. 19980019015. Stability Analysis of Plates and Shells; UNITED STATES; 1998

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“Optimal design of general stiffened composite circular cylinders for global buckling with strength constraints”, Composite Structures, Vol. 41, Nos. 3-4, March-April 1998, pp. 243-252, doi:10.1016/S0263-8223(98)00020-8

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“Optimal design of grid-stiffened composite panels using global and local buckling analyses”, Journal of Aircraft, 1998, Vol. 35, No. 3, pp. 478-486, presented at AIAA 37th SDM Conference, Salt Lake City, UT

Navin Jaunky (1), Norman F. Knight, Jr. (1), and Damodar R. Ambur (2)

(1) Department of Aerospace Engineering, Old Dominion University, Norfolk, VA 23529-0247, USA

(2) Structural Mechanics Branch, NASA Langley Research Center, Hampton, VA 23681-0001, USA

“Buckling analysis of anisotropic variable-curvature panels and shells”, Composite Structures, Vol. 43, No. 4, December 1998, pp. 321-329, doi:10.1016/S0263-8223(98)00118-4

Norman F. Knight, Jr. (Aerospace Engineering Department, Old Dominion University, Norfolk, VA 23529-0247), “Chapter 1 Finite element techniques for nonlinear postbuckling and collapse of elastic structures”, in Finite element analysis (FEA) techniques, edited by Cornelius T. Leondes, Structural Dynamic Systems Computational.... (1998) (cannot copy ABSTRACT and reference are not shown on website)

Bushnell, David (1), Jiang, Hao (2) and Knight, Norm F., Jr. (2)

(1) Lockheed Martin Advanced Technology Center, Palo Alto, CA

(2) Old Dominion Univ., Norfolk, VA

“Additional buckling solutions in PANDA2”, AIAA-1999-1233, 40th AIAA Structures, Structural Dynamics and Materials Conference, April 1999

Navin Jaunky and Norman F. Knight Jr. (Old Dominion University, Norfolk, VA 23529-0247, USA), “An assessment of shell theories for buckling of circular cylindrical laminated composite panels loaded in axial compression”, International Journal of Solids and Structures, Vol. 36, No. 25, September 1999, pp. 3799-3820, doi:10.1016/S0020-7683(98)00177-2

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“STAGS computational procedure for progressive failure analysis of laminated composite structures”, International Journal of Non-Linear Mechanics, Vol. 37, Nos. 4-5, June 2002, pp. 833-849, Special issue: Stability & Vibration in Thin-Walled Structures, doi:10.1016/S0020-7462(01)00101-9

Norman F. Knight, Jr., and Thomas J. Stone (Veridian Systems Division, Chantilly, Virginia), “Rapid Modeling and Analysis Tools: Evolution, Status, Needs and Directions, NASA/CR-2002-211751, 2002

Norman F. Knight, Jr. (General Dynamics – Advanced Information Systems, Chantilly, Virginia), “Bearing-Load Modeling and Analysis Study for Mechanically Connected Structures”, NASA/CR-2006-214529, NASA Langley Research Center, December 2006

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“Finite Element Modeling of the Buckling Response of Sandwich Panels”, AIAA 43rd Structures, Structural Dynamics and Materials Conference, AIAA-2002-1517, 2002

Knight Jr.N.F., 2006, User-Defined Material Model for Progressive Failure Analysis, NASA/CR-214526.

Some unpublished words written by Dr. Norm Knight about buckling. Written by Norm in April, 2020:

In times past before too much FEA, structural stability was often decided by testing and then some with analysis. The old SP 8007 approach seemed to be based mostly on testing and compared to “buckling” analyses that were really eigenvalue analyses rather than nonlinear collapse analyses – not positive here but I think this is true. The so-called “buckling analysis” gave eigenvalues that were then related to “buckling loads”; however, they were way off the test “buckling load”. This gave rise to the knockdown factors that accounted for in some way uncertainty in boundary condition, materials, general imperfections (surface waviness and/or thickness variations), and so on. Then the buckling analysis value was “knockdown” by the factor from SP8007. This is more of a 5,000 foot description.

What I am unsure of is what the tests cited as the basis for the SP8007 actually represented? Was it a local buckling event? Was it a global buckling event? Was it the initial drop in load? Or was it a collapse? For plates we generally see a decrease in stiffness (slope) of a load vs end shortening response but the slope stays positive and load can be increased. Not so for shells. So I wonder just what the test data for SP8007 really represented (local “buckles”, global “buckles” or collapse)?

Anyway jump ahead a decade or two and we then started using nonlinear FEA and incorporated geometric imperfections and even used linear combination of the eigenvectors as an assumed imperfection for the structure. Riks’ method came in and we could go further along on some unstable paths with the assumption that the response was all statically connected – no mode jumping.

Now it seems to me, and I may be wrong, that the current generation of analysts have forgotten about buckling eigenvalue analyses and probably courses on structural stability are infrequently offered. When I was in academia, I was only able to populate (4 students!) and teach a buckling course once in 8 years. Fundamental mechanics courses were difficult to populate, my experience. Now it seems detailed FEA models are built and then executed using a combination of implicit and explicit solution schemes. A nonlinear analysis is started statically and when convergence is slow, they switch to a dynamic analysis (perhaps an implicit scheme with

inherent damping or an explicit scheme that only required the internal force vector and no a 'tangent' stiffness matrix.

So when the design spec calls for an assessment of "buckling", what does the analyst do? Eigenvalue analysis of a detailed FEA? Standard nonlinear analysis until convergence stops or negative eigenvalues are detected? Or peak load from an implicit/explicit nonlinear analysis? How I have read and understood the standards is that I think it should be an eigenvalue analysis of a perfect structure based on minimum design thicknesses combined with the factors from SP8007. Then we augment our understanding with additional analysis. On the other hand, maybe it is like the effort that Mark Hillburger and company have been doing to assess what it means. To me, the old SP8007 type of approach really "protected" us from some of the "unknown" unknowns and as we drive out the "known" unknowns ((e.g., load introduction issues, boundary effects, imperfections) we increase risk.